

# Condor-E Small Spacecraft with an electrooptical sensor

# **ADVANTAGES**

- Resolution of EOS in panchromatic band > 1 m
- Possible stereophotography in one orbit
- Mass of spacecraft up to 1150 kg
- Possible re-aiming Of EOS observing line in roll and pitch
- Application of common space platform

#### **DESIGNATION**

CONDOR-E small spacecraft with an electro-optical sensor is designed for collection, storage and transmission of earth remote sensing data in visible and infra red bands of electromagnetic spectrum to ground data receiving and processing posts. The SSC optical module can be equipped with various sensors operating within panchromatic, multispectrum, near and medium IR band.

## **COMPOSITION**

Composition of Condor-E SSC with EOS:

- common space platform (CSP);
  - onboard control complex (OCC);
  - data collection system (DCS);
- data transmission system (DTS);
- propulsion plant (PP);
- air conditioning system (ACS);
- power generation system (PGS);
- electro-optical camera (EOC).

SPECIFICATIONS OF CONDOR-E SSC	WITH	E0:
Parameters of operation orbit for SSC:		
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altitude     inclination	$\sim 500~\text{km}$ up to $98^\circ$
SSC mass	up 1,150 kg
EOS mass	up 350 kg
Data transmission rate	up to 350 Mbit/sec
DTS frequency band	X-band
SSC active life	5 years

EOS SPECIFICATIONS				
Spectral bands	Visible, IR			
Coverage	1,000 km			
Swath	> 12 km			
Resolution in visible band	≥ 1 m			
Resolution in IR band	5-10 м			
Turn of EOS: • in orbital plane • in roll	± 30° ± 45°			



## SCHEME OF STRIP SURVEY

### SCHEME OF CONVERGENT SURVEY/STEREOPHOTIGRAPHY





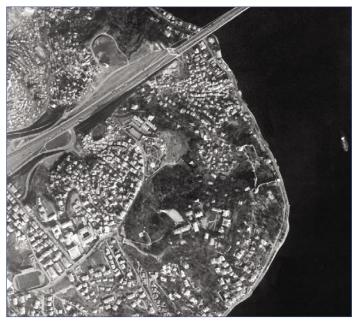
## **OPTICAL IMAGE SAMPLES**

#### KVR-1000



High resolution panchromatic image, Bangkok region

### KVR-1000



High resolution panchromatic image of coastal area, Istanbul



Panchromatic, high resolution image, San-Francisco, USA

